



The Living With a Star (LWS) Sentinels Mission

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Sentinels STDT



Robert P. Lin (Chair)	UCB	Spiro K. Antiochos	NRL
Stuart D. Bale	UCB	Joseph M Davila	GSFC
Antoinette B. Galvin	UNH	Dennis K. Haggerty	APL
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Richard A. Mewaldt	Caltech	Neil Murphy	JPL
Geoff D. Reeves	LANL	Pete Riley	SAIC
James M. Ryan	UNH	Karel Schrijver	Lockheed
Rainer Schwenn	MPI Lindau	Allan J. Tylka	NRL
Thomas Zurbuchen	U Mich		
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Ex-Officio and other non-members:

Adam Szabo	GSFC	Sentinels Study Scientist
Michael Wargo	NASA/HQ	Exploration Representative
Lika Guhathakurta	NASA/HQ	Program Scientist
Chris StCyr	GSFC	LWS Sr. Project Scientist
Haydee M. Maldonado	GSFC	Project Manager
Hermann Opgenoorth	ESA	ILWS Chair
Ronald D Zwickl	NOAA/SEC	User Community Representative



Sentinels STDT

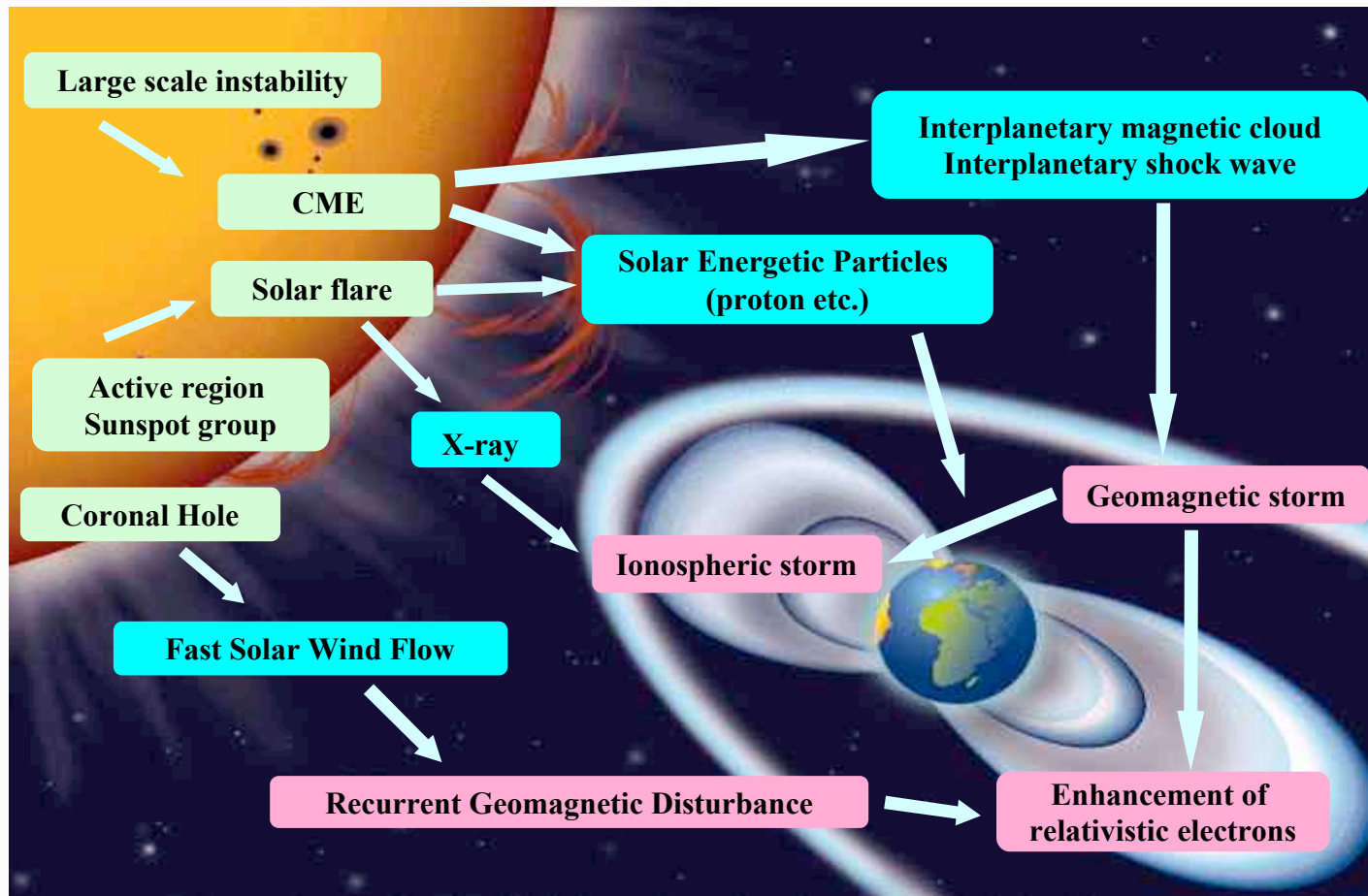


STDT Meetings:

Full Meeting #1	Sept 8-10, 2004	Berkeley
Full Meeting #2	Feb 2-4, 2005	Berkeley
Full Meeting #3	Apr 11-13, 2005	APL
Full Meeting #4	Jun 29 - Jul 1, 2005	Berkeley
Full Meeting #5	Sept 7-9, 2005	U. Michigan
Writers' Meeting	Oct 16-20, 2005	Wintergreen
Writers' Meeting	Mar 2-3, 2006	Greenbelt
Report Release	May 24, 2006	Spring AGU



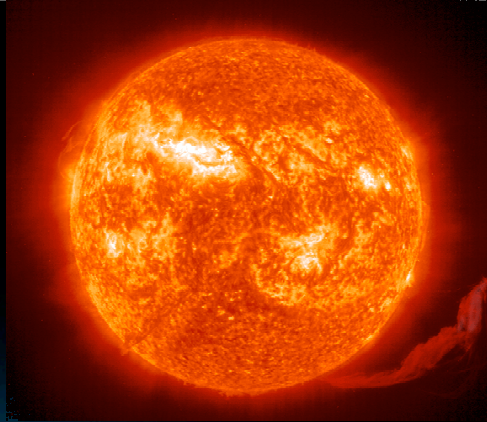
Sentinels Primary Objective



Discover, understand and model the heliospheric initiation, propagation and solar connection of those energetic phenomena that adversely affect space exploration and life and society here on Earth.

Sentinels Science Objectives

- Determine where, when and how are solar energetic particles accelerated and their transport.
- Determine the origin, evolution and interaction of CMEs, shocks and other transient solar wind structures.
- Characterize the interplanetary environment (worse case scenarios)
- Develop forecasting capabilities for Earth, Mars and for spacecraft in transit.





SEP Focused Science Questions



- Determine the roles of CME-driven shocks, flares and other processes in accelerating energetic particles.
 - _ When and where are energetic particles accelerated by the Sun?
 - _ How are the energetic particles observed at the Sun related to those observed in the interplanetary medium?
 - _ What conditions lead to the jets associated with impulsive SEP events?
 - _ What physical processes accelerate SEPs?
- Identify the conditions that determine when CME-driven shocks accelerate particles to high energy.
 - _ What are the seed populations for shock-accelerated SEPs and how do they affect SEP properties?
 - _ How do CME/shock structure and geometry as well as ambient conditions affect SEP acceleration?
- Determine how energetic particles are transported from their acceleration site and distributed in radius, longitude and time.
 - _ What processes scatter and diffuse SEPs both parallel and perpendicular to the magnetic field?
 - _ What are the relative roles of scattering, solar wind convection and adiabatic cooling in SEP event decay?



CME Focused Science Questions



- Determine the physical mechanisms of eruptive events that produce SEPs.
 - _ What solar conditions lead to the initiation of a fast CME?
 - _ How does the pre-eruption corona determine the SEP-effectiveness of a CME?
 - _ How close to the Sun and under what conditions do shocks form?
- Determine the multiscale plasma and magnetic properties of ICMEs and shocks.
 - _ How does the global 3D shape of ICMEs/shocks evolve in the inner heliosphere?
 - _ How does CME structure observed at the Sun map into the properties of interplanetary CMEs?
- Determine how the dynamic inner heliosphere influences the evolution of ICMEs.
 - _ How is the inner heliospheric solar wind determined by coronal and photospheric structure?
 - _ How do ICMEs interact with the preexisting heliosphere?
 - _ How do ICMEs interact with each other?



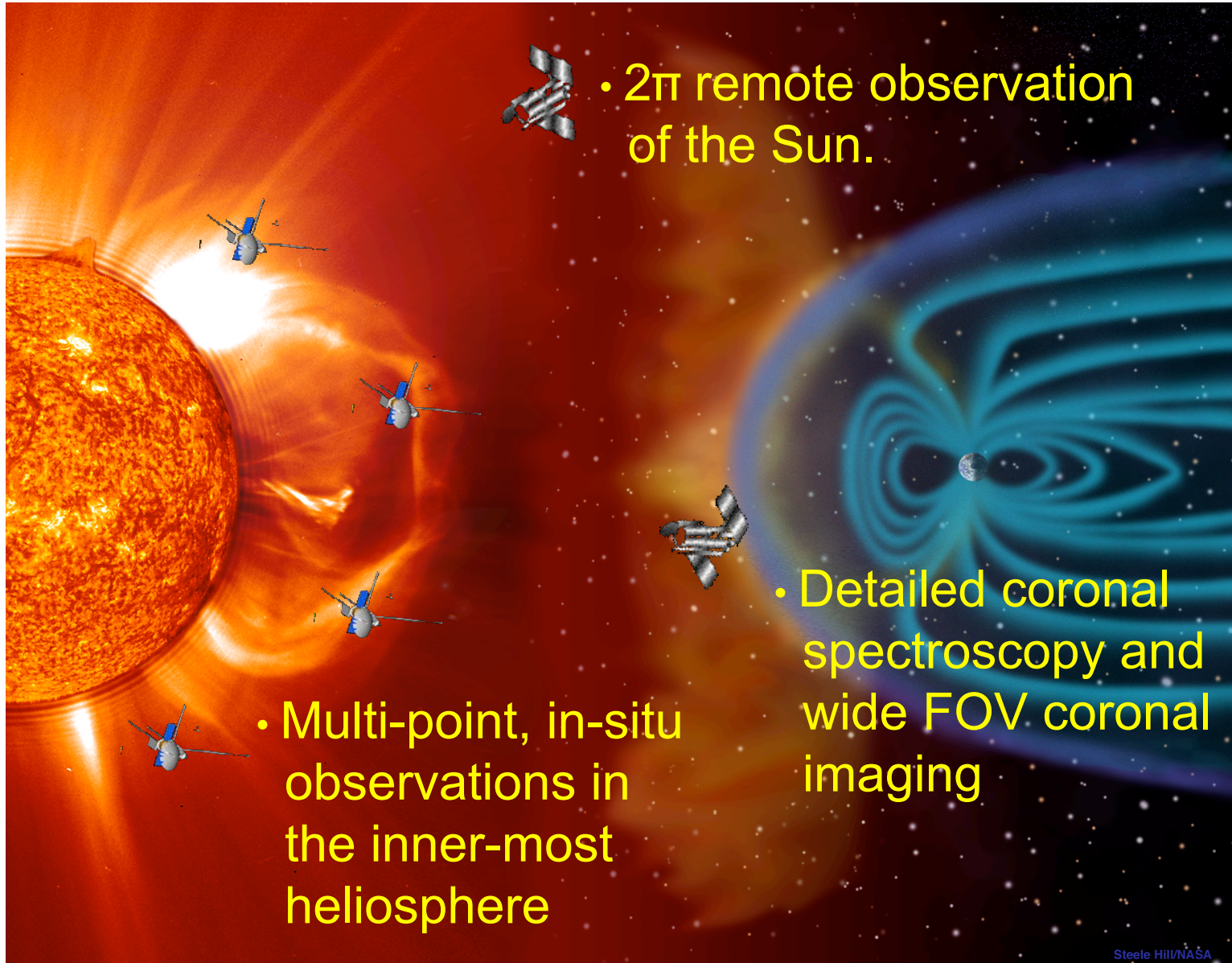
Sentinels Modeling Requirements



- Global Heliospheric Models
 - Inputs: 2π photospheric magnetic maps
coronal plasma conditions
 - Validation: dispersed inner heliospheric in-situ observations
 - Enhancements: data assimilation
- Transient Dynamics Models (e.g., CME onset and evolution)
 - Inputs: high resolution photospheric vector magnetic fields
coronal plasma conditions
 2π photospheric magnetic maps
 - Validation: multi-point in-situ measurements of CMEs in inner heliosphere
coronal plasma diagnostics
white-light coronagraph images of structures
 - Enhancements: data assimilation
- SEP Acceleration and Transport Models
 - Inputs: turbulence close to the Sun
source population and SEP properties near acceleration site
plasma and magnetic configuration at CME-driven shocks
 - Verification: in-situ SEP observations at different longitudes
 - Enhancements: combining global, initiation and particle codes



Sentinels Measurement Requirements





Sentinels Observational Strategy

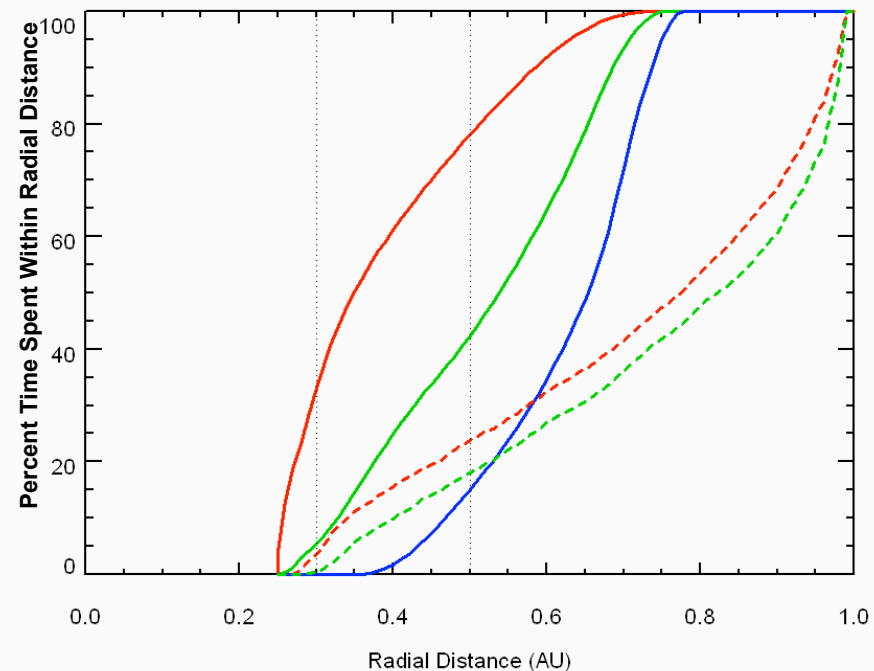


• Inner Heliospheric in-situ Observations

- Close to the Sun: 1-2 SEP mean-free-paths (< 0.3 AU)
- Sufficient duration: 10s of SEP events ($> 30\%$ duty cycle below 0.3 AU)
- # of points: Minimum of 4 s/c for CME geometry and SEP field line connection

• In-situ and Imaging Observation Overlap

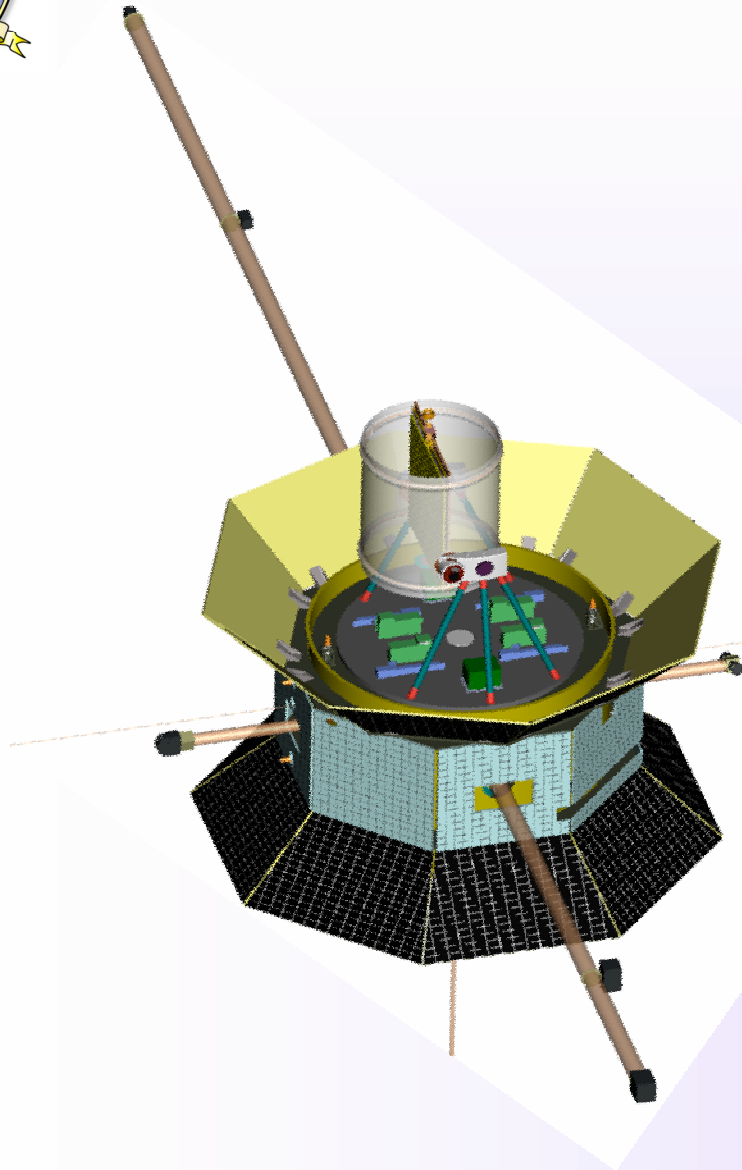
- Duration: > 1 year
- FOV: Coronagraph FOV
 ~ 0.3 AU (60 Rs)



— 1 Sentinel — 2 Sentinels — 3 Sentinels - - 1 Helios - - 2 Helios



Inner Heliospheric Sentinels



Instruments

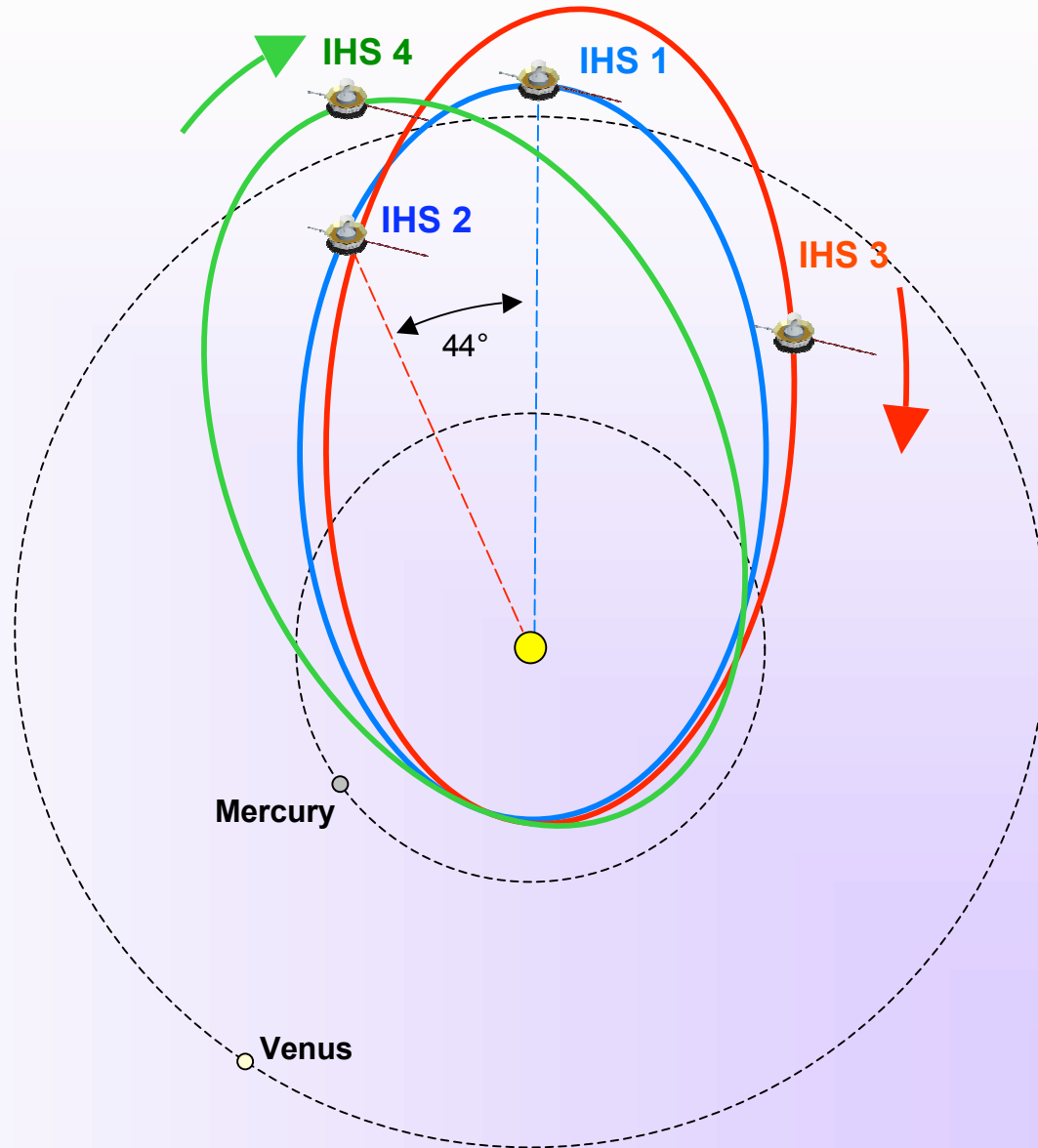
Dual Magnetometer
AC Magnetic Fields Search Coil
Radio Science
Solar Wind Ions
Solar Wind Electrons
Solar Wind Composition
Suprathermal Electrons and Ions
Low Energy Ions and Electrons
High Energy Ions and Electrons
SEP Charge State
Neutron Spectrometer
X-Ray Imager
Gamma-Ray Spectrometer



Inner Heliospheric Orbit Design



- 3 Venus gravity assists for each spacecraft
- Final orbits:
0.25 x 0.76 AU
- Orbital periods:
127-137 days
- Cruise:
2 yr 3-11 months
- Launch opportunities:
March 2012, Feb 2014,
Sept 2015, March 2017





Inner Heliospheric Sentinels Spacecraft Design



- 4 identical spin stabilized spacecraft
- Spin axis: Ecliptic North
- Launch vehicle: single Atlas V-541
- C_3 : 23-27 km²/s² depending on launch opportunity
- Delta V: 100 m/s per s/c

- Spacecraft dry mass: 504 kg per spacecraft
- Instrument mass: 70.5 kg per spacecraft
- Total launch mass with margins: 3192 kg

- Power generated at 1 AU: 220 W
- Power generated at 0.25 AU: > 500 W
- Radiation tolerance: 6 krad

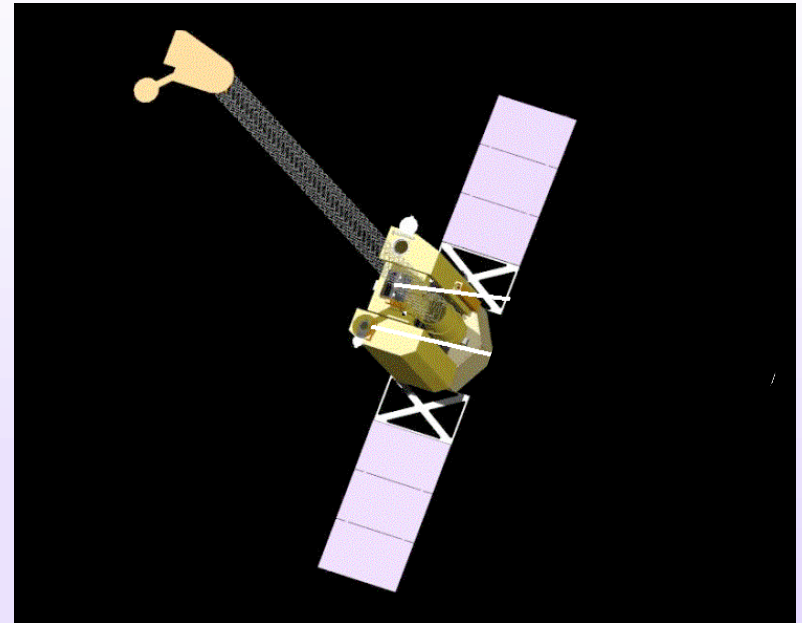
- Telemetry rate: 6.5 kbps
- RF frequency: X-band
- RF transmit power: 100 W max
- Real time telemetry



Near-Earth Imaging Sentinel



- Sun-sync Earth orbit.
- Significant overlap with IHS and SDO.
- Instrumentation:
 - Inner Coronagraph (1.3 – 5 Rs)
 - Outer Coronagraph (4 – 55 Rs)
 - UV Spectroscope
- Similar concept previously proposed under Midex.





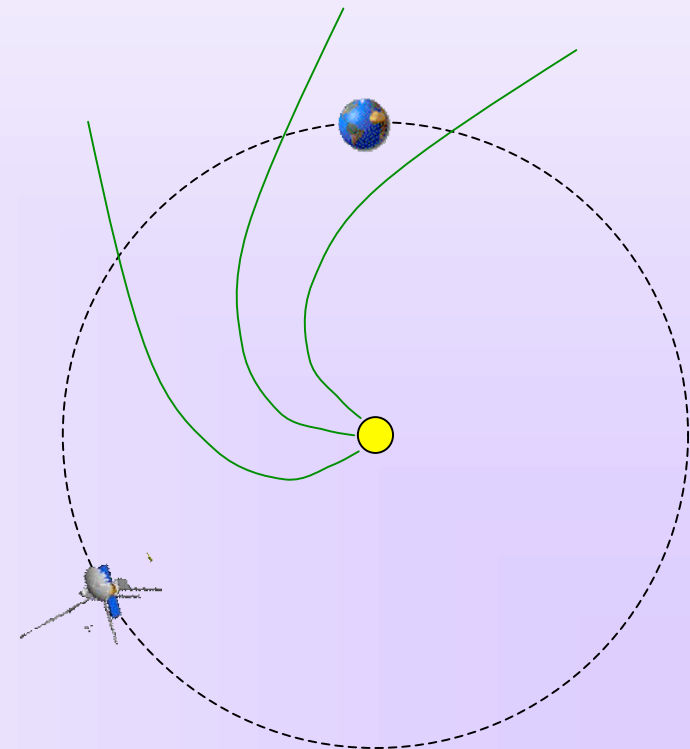
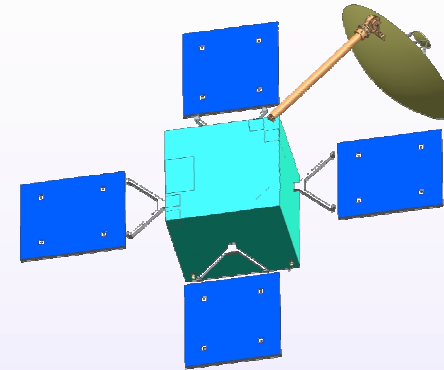
Far Side Sentinel



- 1 AU orbit, 120° - 180° leading Earth.
- Taurus launch.
- Total launch mass: 250-350 kg
- Significant overlap with IHS and Solar Orbiter.
- Instrumentation:

Full Disk Magnetograph

Optional small in-situ package



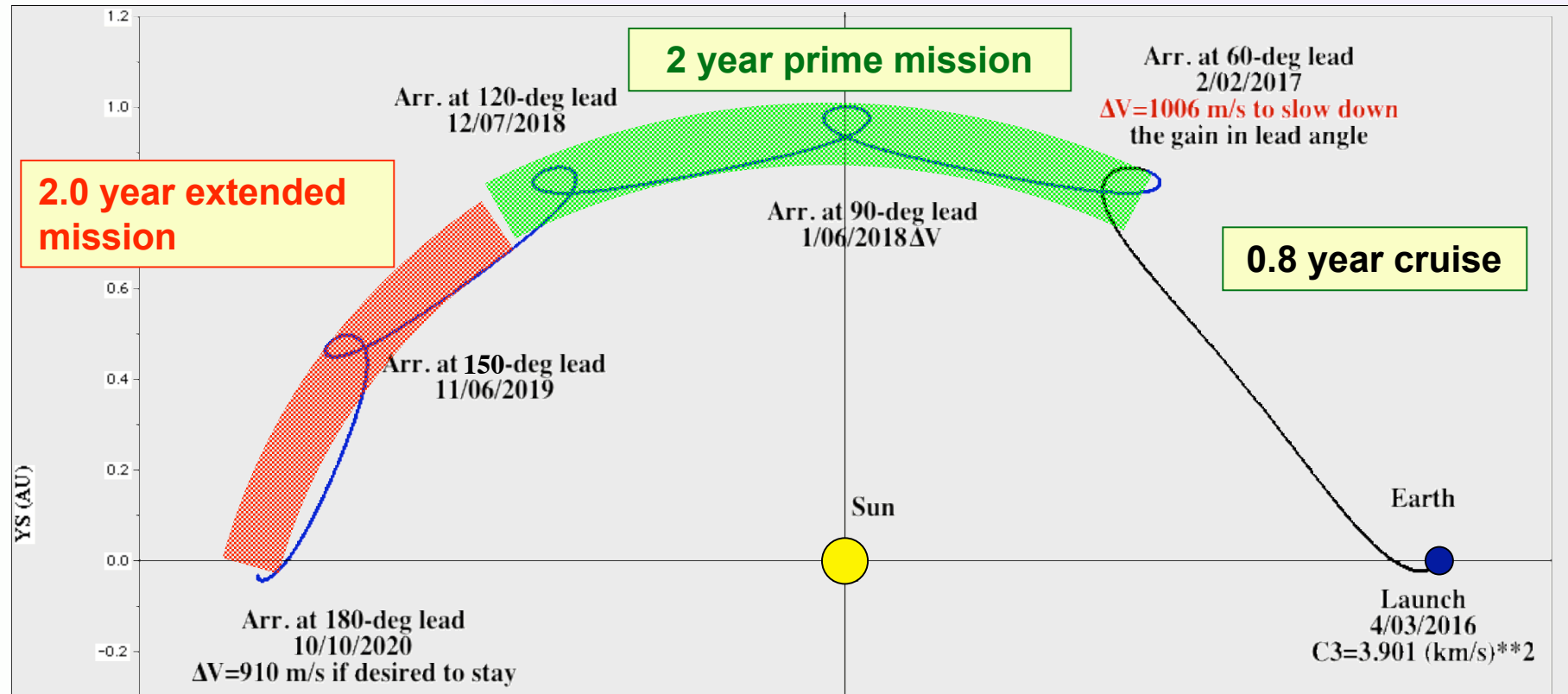


Far Side Sentinel



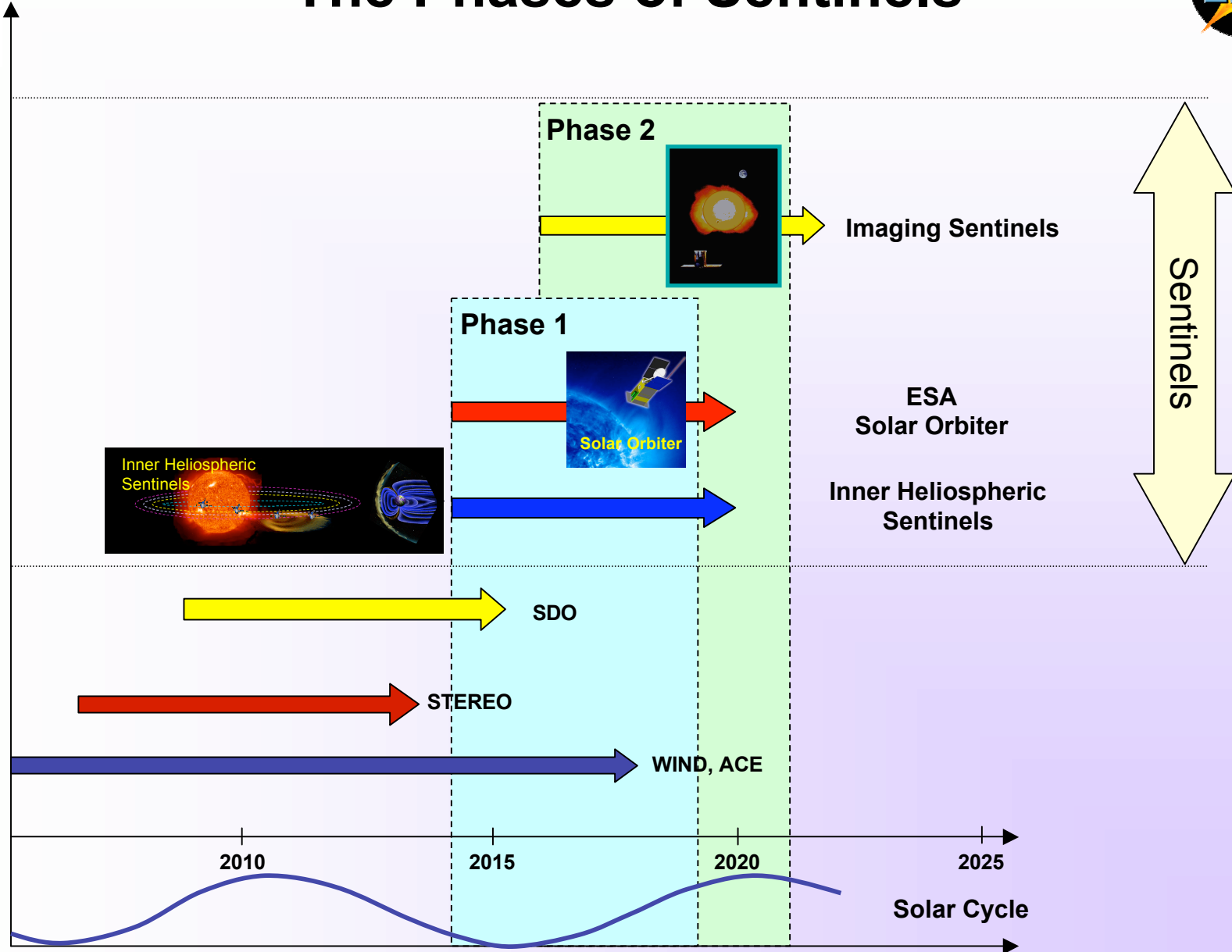
Ballistic trajectory that minimizes time to 60 degrees and then drifts from 60 to 180 degrees in < 4 years

- $C3 = 3.9\text{-}4.5 \text{ km}^2/\text{s}^2$
- $\Delta V = 85 \text{ m/s}$
- Launch Vehicle: Taurus 2130





The Phases of Sentinels





ESA Solar Orbiter and Sentinels



- Inner heliospheric (0.22 x 0.9 AU) mission in the same time frame as IHS.
- Both in-situ and remote sensing instrumentation.
- 2nd half of mission to latitudes above 30°.

